## NASA/CR-2001-210952



# Short-Term Aging of NeFeB Magnets for Stirling Linear Alternator Applications

Janis M. Niedra Dynacs Engineering Company, Inc., Brook Park, Ohio Since its founding, NASA has been dedicated to the advancement of aeronautics and space science. The NASA Scientific and Technical Information (STI) Program Office plays a key part in helping NASA maintain this important role.

The NASA STI Program Office is operated by Langley Research Center, the Lead Center for NASA's scientific and technical information. The NASA STI Program Office provides access to the NASA STI Database, the largest collection of aeronautical and space science STI in the world. The Program Office is also NASA's institutional mechanism for disseminating the results of its research and development activities. These results are published by NASA in the NASA STI Report Series, which includes the following report types:

- TECHNICAL PUBLICATION. Reports of completed research or a major significant phase of research that present the results of NASA programs and include extensive data or theoretical analysis. Includes compilations of significant scientific and technical data and information deemed to be of continuing reference value. NASA's counterpart of peerreviewed formal professional papers but has less stringent limitations on manuscript length and extent of graphic presentations.
- TECHNICAL MEMORANDUM. Scientific and technical findings that are preliminary or of specialized interest, e.g., quick release reports, working papers, and bibliographies that contain minimal annotation. Does not contain extensive analysis.
- CONTRACTOR REPORT. Scientific and technical findings by NASA-sponsored contractors and grantees.

- CONFERENCE PUBLICATION. Collected papers from scientific and technical conferences, symposia, seminars, or other meetings sponsored or cosponsored by NASA.
- SPECIAL PUBLICATION. Scientific, technical, or historical information from NASA programs, projects, and missions, often concerned with subjects having substantial public interest.
- TECHNICAL TRANSLATION. Englishlanguage translations of foreign scientific and technical material pertinent to NASA's mission.

Specialized services that complement the STI Program Office's diverse offerings include creating custom thesauri, building customized data bases, organizing and publishing research results . . . even providing videos.

For more information about the NASA STI Program Office, see the following:

- Access the NASA STI Program Home Page at http://www.sti.nasa.gov
- E-mail your question via the Internet to help@sti.nasa.gov
- Fax your question to the NASA Access Help Desk at 301–621–0134
- Telephone the NASA Access Help Desk at 301–621–0390
- Write to:

NASA Access Help Desk NASA Center for AeroSpace Information 7121 Standard Drive Hanover, MD 21076

## NASA/CR—2001-210952



# Short-Term Aging of NeFeB Magnets for Stirling Linear Alternator Applications

Janis M. Niedra Dynacs Engineering Company, Inc., Brook Park, Ohio

Prepared under Contract NAS3-98008

National Aeronautics and Space Administration

Glenn Research Center

Trade names or manufacturers' names are used in this report for identification only. This usage does not constitute an official endorsement, either expressed or implied, by the National Aeronautics and Space Administration.

Available from

NASA Center for Aerospace Information 7121 Standard Drive Hanover, MD 21076 National Technical Information Service 5285 Port Royal Road Springfield, VA 22100

### SHORT-TERM AGING OF NeFeB MAGNETS FOR STIRLING LINEAR ALTERNATOR APPLICATIONS

Janis M. Niedra
Dynacs Engineering Co., Inc.
Brook Park, Ohio 44142

#### **Summary**

NeFeB type magnets have been proposed for use in free piston Stirling engine driven, linear alternators to generate electric power during long duration space missions. These type of materials provide the highest energy product commercial magnets, thus minimizing alternator size or mass, but do not provide the high temperature stability of magnetic properties found in the SmCo type magnets. Therefore, to apply the NeFeB type magnets at elevated temperatures to multiyear space missions, their long-term aging characteristics must be determined.

This report presents 200 hour aging data for 6 types of NeFeB magnets selected from 3 manufacturers. Aging was performed under vacuum at 150 C, with a steady demagnetizing field of 5 kOe applied. From the data produced by this short-term aging run, candidate magnet types were selected for a planned 12,000 hour long-term run. Depending on the manufacturer's magnet type, remanence losses observed ranged from 0 to 7%, when measured at 120 C on an established recoil line. Also, intrinsic coercivity losses up to about 4% were observed for the M-H curve at 120 C. In some cases, these coercivity losses were not recoverable by recharge of the magnet, indicating a structural change of the material.

#### **Need for Permanent Magnet Aging Data**

To date, the highest permanent magnet energy products,  $(BH)_{max}$ , have been achieved in magnet materials based on the elements neodymium, iron and boron. A variety of commercial magnets is available in this class, differing in magnetic properties such as remanence  $(B_r)$ , intrinsic coercivity  $(_MH_C)$  and temperature capability. These variations are achieved by the addition of other elements, together with generally proprietary processing steps. Hence the magnets can be identified only as belonging to the NeFeB class and the manufacturer's specific designation.

It is well known that the NeFeB type magnets do not have the high temperature stability of magnetic properties possessed by the SmCo type magnets, which can operate reliably at 300 C, or even higher. Therefore, to take advantage of the high energy product (equivalent to a high  $B_r$  for a flat topped M-H curve in the 2nd quadrant) of the NeFeB type magnets at elevated temperatures in long duration space travel applications, the long-term aging characteristics of these magnets must be known. Unfortunately, neither long-term (order of years) nor even short-term (a few hundred hours) magnet aging data is readily available for the commercial magnets.

Final aging data to qualify a magnet type for a long term mission has to be taken long term itself and under bias conditions of temperature and demagnetizing field at least as severe as expected in the application. No reliable and inclusive formula for accelerated aging can be given, due to the possibility of multiple aging mechanisms having various activation thresholds. The short-term tests described below

were done to preselect 2 or 3 NeFeB magnet types from several manufacturers as the ones most likely to give the best performance in a multi thousand hour aging run.

#### **Magnet Selection for the Short-Term Aging Run**

Two types of anisotropic, NeFeB magnet were selected from each of 3 manufacturers:

Vacuum Schmeltze (VAC): 383HR, 396HR Ugimag: 38KC2, 40HC2 Magnequench: MQ3-F36, MQ3-F42.

The magnet selection criteria were primarily  $B_r$  and  $_MH_C$ . A  $B_r$  below 1.2 T at 21 C was deemed to be uninteresting, as 1.2 T is not much above the  $B_r$  of some SmCo type magnets. Likewise, an  $_MH_C$  below about 17 kOe at 21 C was thought to be too low to provide adequate coercivity safety margin at around 100 C. Due to an inverse relationship between  $B_r$  and  $_MH_C$ , these criteria greatly delimit the candidate materials.

To fill the 10 available sample slots of the aging fixture, 2 samples of each of the 2 Ugimag and 2 Magnequench types and only one sample of each of the 2 VAC types were selected. The VAC type 396HR was not favored due to a relatively low B<sub>r</sub> and the VAC type 383HR was also less favored due to relatively large steps in the top part of its M-H curve.

#### **Aging Bias Conditions**

The following aging bias conditions were chosen for the 200 hour run:

Temperature: 150 C Demagnetizing field: 5.0 kOe.

A demagnetizing field of about 5 kOe was shown in a report [1] dealing with permanent magnet excited linear alternator modeling and tuning to be a fairly typical value. Moreover, at 150 C, a 5 kOe demagnetizing field is already quite close to the knee of the M-H curve in the 2nd quadrant.

The 150 C aging temperature was chosen somewhat arbitrarily as a value significantly above 120 C to accelerate aging, but still within the manufacturers' data limits for the magnets. The 120 C value was previously picked, again somewhat arbitrarily, as a long-term aging temperature sufficiently above the expected real use temperature (~80 C) of the magnets to provide an adequate reliability margin. Preliminary aging runs, 120 hours for the two Magnequench samples and 72 hours for the 40HC2, at 120 C and a 6 kOe demagnetizing field showed little or no resolvable aging effects. There is no more rigorous justification for the choice of these bias conditions.

#### **Experimental Setup of the 200-Hour Run**

This short-term aging run was performed on 1-cm cubic magnet samples in vacuum and under the above given bias conditions. The magnet aging fixture, which controls the sample temperature and applies a demagnetizing field, held 10 samples between 4-inch diameter, iron-cobalt alloy pole pieces. This fixture held the magnets in fixed positions, distributed over a pole face so as to minimize any intersample field interference. The manufacturer of this fixture, KJS Associates of Magnetic Instrumentation, Inc., specified the demagnetizing field uniformity to be within 5% over the pole faces. The source of the applied field was a GMW, Inc. 4-inch electromagnet, energized by a Kepco BOP 20-20M bipolar

operational power supply. This power supply provided a constant 6.58 A current to the electromagnet, with a stability of better than 1 part in 600 over the period of the run. A 3 part in 500 decrease from the room temperature value of the demagnetizing field was observed, as the electromagnet coils and parts of its frame and poles warmed up during the run, but no correction for this small change was attempted.

The vacuum quality and residual gases in the aging fixture were not well determined. At start, with initialized magnet samples in place, the aging fixture was turbopumped for several days at room temperature. Then the temperature was gradually raised to 120 C, while pumping. After about 2 days at 120 C, the pressure at the turbopump entrance dropped to about  $5x10^{-8}$  mm Hg. Clearly, the pressure could have been 10 to 100 times higher in the chamber of the fixture, due to the small pipe leading into the fixture. The prolonged gas load from the fixture may well have come from outgassing at temperature of its Viton o-rings or even from the possibly porous magnet samples, as no helium leak could be detected. Relative to the subsequent aging conditions, this 120 C bakeout was quite harmless, because the magnets then were subjected to a uniform self demagnetizing field of less than 1 kOe.

#### **Magnet Initialization and Measurements**

To get meaningful aging data, the magnets need to be "stabilized", or initialized, on a well defined M-H recoil line defined by the bias conditions during aging. It has been shown by machine computation [2] of the self-demagnetizing field of a 1-cm cubic magnet in free space, that at 150 C this field is sufficient to cause local demagnetization of a fully charged sample of the type measured here. However, taking of the M-H curve data requires that a magnet sample be briefly exposed to free space in order to reset electronic integrators. This means that the M-H curve aging data has to be taken at a temperature sufficiently lower to avoid the self-demagnetization danger. This temperature was picked to be 120 C, as it is sufficiently low to avoid the danger and is the same as the temperature of the planned long-term aging runs.

Accordingly, the magnets were initialized at 150 C by repeated application (back and forth on recoil line) of a demagnetizing field up to 5.0 kOe in the aging fixture and then cooled at zero field while still in the fixture. This establishes a recoil line at 150 C. And it also induces corresponding recoil lines, but of unknown field amplitude, at other temperatures. Thus at a lower temperature, a larger demagnetization field can be applied without disturbing the established recoil line. In this way, the  $B_r$  on the recoil line could be measured at 120 C, before and after the aging run, to determine the fractional change  $\Delta B_r/B_r$ .

Measurement of the intrinsic coercivity aging  $\Delta_M H_C/_M H_C$  requires taking the full M-H demagnetization curve, which obviously erases the magnetization history. Hence this data curve can only be taken once, at say 120 C.

#### **Experimental Results**

The data from the 200-hour aging run at 150 C and -5.0 kOe is reported in Table I. The important magnetization loss data is the decrease in remanence  $B_r$ , measured before and after aging on the established recoil line. And the important measure of loss in resistance to demagnetization is the decrease in intrinsic coercivity  $_MH_C$ , measured before and after aging on the M-H curve. Hence included are the  $B_r$  and  $_MH_C$  on the initial M-H curve and the  $B_r$  and  $_MH_C$  on the M-H curve of the sample recharged after aging. A decrease in this latter  $B_r$  indicates a permanent loss of magnetic moment due to a metallurgical change. Non-recovery of the full coercivity  $_MH_C$  after recharge of an aged sample likewise indicates a basic structural change.

Description of columns in Table I:

- 1. The remanence B<sub>r</sub> before aging, measured on the saturated M-H curve.
- 2. The remanence B<sub>r</sub> before aging, measured on the recoil line.
- 3. The remanence  $B_r$  after aging, measured on the recoil line.
- 4. The remanence B<sub>r</sub> of a sample recharged after aging, measured on the saturated M-H curve.
- 5. The intrinsic coercivity MH<sub>C</sub> before aging, measured on the saturated M-H curve.
- 6. The intrinsic coercivity MHC after aging, measured on the M-H curve.
- 7. The intrinsic coercivity MH<sub>C</sub> of a sample recharged after aging, measured on the saturated M-H curve.

Table II presents the fractional aging of the remanence and coercivity, calculated from the data in Table I. The initial  $B_r$  and  $_MH_C$  data is also repeated for reference. It can be seen that many of these fractional losses are at the 1 to 2 percent level, with 1% being close to the resolution limit of the experimental apparatus. Some of the losses are not resolvable from zero.

Description of columns in Table II:

- 1. The remanence  $B_r$  before aging, measured on the saturated M-H curve. Repeats Column (1) of Table I
- 2. The remanence B<sub>r</sub> before aging, measured on the recoil line. Repeats Column (2) of Table I.
- 3. Fractional loss of remanence  $B_r$  of a sample recharged after aging, measured on the saturated M-H curves.  $\Delta B_r \equiv B_{r, \text{ final}} B_{r, \text{ initial}}$ .
- 4. Fractional loss of remanence  $B_r$ , measured on the recoil line.  $\Delta B_{r, \text{ recoil}} \equiv B_{r, \text{ final, recoil}} B_{r, \text{ initial, recoil}}$ .
- 5. The intrinsic coercivity  $_MH_C$  before aging, measured on the saturated M-H curve. Repeats Column (5) of Table I.
- 6. Fractional loss of intrinsic coercivity  $_MH_C$ .  $\Delta_MH_c \equiv _MH_{c, final} _MH_{c, initial}$ , where  $_MH_{c, final}$  is measured on the "aged" M-H curve.
- 7. Fractional loss of intrinsic coercivity  $_MH_C$ .  $\Delta_MH_c \equiv _MH_{c, \, recharged} _MH_{c, \, initial}$ , where  $_MH_{c, \, recharged}$  is measured on the saturated M-H curve of the sample recharged after aging.

With regard to  $\Delta B_r/B_r$  on the recoil line at 120 C, the 10 samples fit into the following groups:

< 1% loss: 396HR, 40HC2

~ 1% loss: 38KC2

≥ 2% loss: 383HR, MQ3-F36, MQ3-F42.

Thus there seems to be a vague inverse correlation between the  $(\Delta B_r/B_r)_{recoil}$  at 120 C and the  $_MH_C$  at 21 C, as inspection of columns 4 and 5 of Table II shows. Only the MQ3-F42 and the 383HR indicated a small (~1%), non-recoverable loss of magnetic moment, which, however, was temperature dependent.

Table II clearly shows a potentially serious, but less discussed phenomenon, namely that the intrinsic coercivity  $_{\rm M}H_{\rm C}$  also ages. On the 120 C aged M-H curve, this loss in coercivity amounted to about 1 to 2 % for all samples except for the MQ3-F42, which suffered a loss over 3%. This loss tended to persist (with altered values) for all samples even after recharge, indicating a structural change that affects domain wall pinning.

#### **Conclusions and Discussion**

A first cut at selecting magnet types from the set discussed in this report would be to eliminate both of the Magnequench samples, as they exhibited a 3 to 7 % loss of magnetization when measured at 120 C, after being aged for 200 hours at 150 C, with a 5.0 kOe demagnetizing field applied. The remaining candidates are the two VAC types and the two Ugimag types. The VAC type 396HR seems, however, uninteresting, because its 1.20 T remanence (B<sub>r</sub>) at 21 C is the lowest among the samples and not far above that achievable with the more temperature stable SmCo type materials. This pares the candidates down to the VAC 383HR and the Ugimag 40HC2 and 38KC2. At least from the data, the VAC type 383HR does not seem to have anything going in its favor over the Ugimag types. In fact, the 383HR appears to have a twice as high rate of loss of magnetization, compared to the 40HC2 and 38KC2. The remaining 2 magnet types are unfortunately from the same manufacturer. Their loss of remanence was less than 1% and loss of intrinsic coercivity averaged about 2%.

Restriction to just 2 magnet types for the long-term aging run is acceptable, as that allows 5 samples for each type in the present 10-sample aging fixture. Aging more than 2 types simultaneously would make for sparse data for at least one of the types. Sample type distributions such as 2-4-4 or 1-4-5 are of course feasible, but in the group studied here, there is no sample of sufficient interest for the intended application to justify giving it a seat in the fixture.

As observed for the Ugimag samples, the aging data hints at an irreversible increase in top slope and knee rounding of the M-H demagnetization curve. If indeed progressive, this additional effect may lead to accelerated aging such that in the long run the Ugimag materials may lose their initial short-term aging advantage. However, in the proposed schedule for sample testing at 200, 1000, 2000, 6000 and 12000 hours, there is an opportunity to alter sample selection at say the 2000 hour point, with a relatively small loss in time.

#### References

- 1. J.M. Niedra, "Lightweight Linear Alternators With and Without Capacitive Tuning", NASA CR-185273, June 1993.
- 2. Steven M. Geng, private communication, NASA Glenn Research Center.

Aging environment: Period: 200 hours; Aging temperature: 150 C Demagnetizing H-field: 5.0 kOe; Recoil line established at 150 C and  $H_{demag}$  = 5.0 kOe; X: no data possible.

	(1)	(2)	(3)	(4)	(5)	(6)	(7)
	B <sub>r</sub> initial, (T)	B <sub>r</sub> initial, on recoil line, (T)	B <sub>r</sub> final, on recoil line, (T)	B <sub>r</sub> after recharge, (T)	<sub>M</sub> H <sub>c</sub> initial, (kOe)	<sub>M</sub> H <sub>c</sub> final, (kOe)	<sub>M</sub> H <sub>c</sub> after recharge, (kOe)
VAC							
396 HR (#1)							
21 C	1.20			1.20	25.1	X	24.8
120 C	1.08	1.06	1.06	1.08	10.8	10.75	10.8
150 C	1.02	1.02	X		7.25	X	
383 HR (#2)							
21 C	1.27			1.26	21.1	X	20.8
120 C	1.12	1.10	1.08	1.12	8.50	8.35	8.50
150C	1.08	1.05	X		5.90	X	
UGIMAG:							
40HC2 (#3)							
21 C	1.28			1.28	20.0	X	19.75
120 C	1.16	1.15	1.145	1.16	8.65	8.55	8.65
150 C	1.11	1.09	X		6.15	X	
40HC2 (#4)							
21 C	1.28			1.28	20.1	X	19.8
120 C	1.16	1.15	1.15	1.16	8.75	8.52	8.45
150 C	1.11	1.10	X		6.20	X	

## TABLE I CONTINUED

	(1)	(2)	(3)	(4)	(5)	(6)	(7)
	B <sub>r</sub> initial, (T)	B <sub>r</sub> initial on	B <sub>r</sub> final on	B <sub>r</sub> after	<sub>M</sub> H <sub>c</sub> initial,	<sub>M</sub> H <sub>c</sub> final, (kOe)	<sub>M</sub> H <sub>c</sub> after
		recoil line, (T)	recoil line, (T)	recharge, (T)	(kOe)		recharge, (kOe)
<u>UGIMAG</u>							
38KC2 (#5)							
21 C	1.28			1.28	20.5	X	20.2
120 C	1.16	1.15	1.14	1.16	8.75	8.60	8.80
150 C	1.11	1.10	X		6.20	X	
38KC2 (#6)							
21 C	1.28			1.28	20.4	X	20.1
120 C	1.16	1.15	1.14	1.16	8.65	8.50	8.60
150C	1.10	1.10	X	1.10	6.08	X	0.00
1300	1.10	1.10	Λ		0.08	Λ	
MAGNEQ.:							
MQ3-F42 (#7)							
21 C	1.30			1.30	17.8	X	17.65
120 C	1.19	1.18	1.125	1.18	8.75	8.45	8.50
150 C	1.14	1.13	X		6.55	X	
MQ3-F42 (#8)							
21 C	1.28			1.28	18.0	X	17.6
120 C	1.17	1.16	1.125	1.16	8.90	8.55	8.55
150 C	1.12	1.10	X		6.60	X	

## TABLE I CONCLUDED

_	(1)	(2)	(3)	(4)	(5)	(6)	(7)
	B <sub>r</sub> initial, (T)	B <sub>r</sub> initial on recoil line, (T)	B <sub>r</sub> final on recoil line, (T)	B <sub>r</sub> after recharge, (T)	<sub>M</sub> H <sub>c</sub> initial, (kOe)	<sub>M</sub> H <sub>c</sub> final, (kOe)	<sub>M</sub> H <sub>c</sub> after recharge, (kOe)
MAGNEQ.:			recon mie, (1)	recharge, (1)	(ROC)		reenarge, (ROC)
MQ3-F36 (#9)							
21 C	1.26			1.26	17.4	X	17.0
120 C	1.15	1.14	1.06	1.15	8.38	8.25	8.25
150 C	1.11	1.10	X		6.50	X	
MQ3-F36 (#10)							
21 C	1.24			1.24	17.45	X	17.05
120 C	1.13	1.12	1.04	1.13	8.40	8.25	8.35
150 C	1.10	1.08	X		6.75	X	

Aging environment: Period: 200 hours; Aging temperature: 150 C Demagnetizing H-field: 5.0 kOe; Recoil line established at 150 C and  $H_{demag}$  = 5.0 kOe; X: no data possible.

	(1)	(2)	(3)	(4)	(5)	(6)	(7)
	B <sub>r</sub> initial, (T)	B <sub>r</sub> initial on	$(\Delta B_r/B_r)$	$(\Delta B_r/B_r)_{recoil}$	<sub>M</sub> H <sub>c</sub> initial,	$(\Delta_{\rm M} H_{\rm c}/_{\rm M} H_{\rm c})$	$(\Delta_{\rm M} H_{\rm c}/_{\rm M} H_{\rm c})$
		recoil line, (T)	after recharge	(on recoil line)	(kOe)		after recharge
<u>VAC</u>							
396 HR (#1)							
21 C	1.20		0.00		25.1	X	-0.012
120 C	1.08	1.06	0.00	0.00	10.8	-0.005	0.00
150 C	1.02	1.02		X	7.25	X	
383 HR (#2)							
21 C	1.27	1.23	-0.008	-0.016	21.1	X	-0.014
120 C	1.12	1.10	0.00	-0.018	8.50	-0.018	0.00
150C	1.08	1.05		X	5.90	X	
<u>UGIMAG:</u>							
40HC2 (#3)							
21 C	1.28		0.00		20.0	X	-0.0125
120 C	1.16	1.15	0.00	-0.004	8.65	-0.012	0.00
150 C	1.11	1.09		X	6.15	X	
40HC2 (#4)							
21 C	1.28		0.00		20.1	X	-0.015
120 C	1.16	1.15	0.00	0.00	8.75	-0.026	-0.034
150 C	1.11	1.10		X	6.20	X	

### TABLE II CONTINUED

	(1)	(2)	(3)	(4)	(5)	(6)	(7)
	B <sub>r</sub> initial, (T)	B <sub>r</sub> initial on recoil line, (T)	$(\Delta B_r/B_r)$ after recharge	$(\Delta B_r/B_r)_{recoil}$ (on recoil line)	<sub>M</sub> H <sub>c</sub> initial, (kOe)	$(\Delta_{\rm M} H_{\rm c}/_{\rm M} H_{\rm c})$	$(\Delta_{\rm M} H_{\rm c}/_{\rm M} H_{\rm c})$ after recharge
UGIMAG							
38KC2 (#5)							
21 C	1.28		0.00		20.5	X	-0.015
120 C	1.16	1.15	0.00	-0.0087	8.75	-0.017	+0.006
150 C	1.11	1.10		X	6.20	X	
38KC2 (#6)							
21 C	1.28		0.00		20.4	X	-0.015
120 C	1.16	1.15	0.00	-0.0087	8.65	-0.017	-0.006
150C	1.10	1.10		X	6.08	X	
MAGNEQ.:							
MQ3-F42 (#7)							
21 C	1.30		0.00		17.8	X	-0.008
120 C	1.19	1.18	-0.008	-0.0466	8.75	-0.034	-0.029
150 C	1.14	1.13		X	6.55	X	
MQ3-F42 (#8)							
21 C	1.28		0.00		18.0	X	-0.022
120 C	1.17	1.16	-0.009	-0.030	8.90	-0.039	-0.039
150 C	1.12	1.10		X	6.60	X	

## TABLE II CONCLUDED

_	(1)	(2)	(3)	(4)	(5)	(6)	(7)
	B <sub>r</sub> initial, (T)	B <sub>r</sub> initial on recoil line, (T)	$(\Delta B_r/B_r)$ after recharge	$(\Delta B_r/B_r)_{recoil}$ (on recoil line)	<sub>M</sub> H <sub>c</sub> initial, (kOe)	$(\Delta_{\rm M} H_{\rm c} I_{\rm M} H_{\rm c})$	$(\Delta_{\rm M} H_{\rm c}/_{\rm M} H_{\rm c})$ after recharge
MAGNEQ.:							
MQ3-F36 (#9)							
21 C	1.26	1.24	0.00	-0.065	17.4	X	-0.023
120 C	1.15	1.14	0.00	-0.070	8.38	-0.016	-0.016
150 C	1.11	1.10		X	6.50	X	
MQ3-F36 (#10)							
21 C	1.24		0.00		17.45	X	-0.023
120 C	1.13	1.12	0.00	-0.071	8.40	-0.018	-0.006
150 C	1.10	1.08		X	6.75	X	

#### REPORT DOCUMENTATION PAGE

Form Approved OMB No. 0704-0188

Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing the collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden, to Washington Headquarters Services, Directorate for Information Operations and Reports, 1215 Jefferson Davis Highlyway, Suite 1204, Arlington, VA 22202-4302, and to the Office of Management and Burdent Paperwork Reduction Project (10704-0188) Washington, DC 20503

Davis Highway, Suite 1204, Annigton, VA 22202-	, ,	•	, , , , , , , , , , , , , , , , , , , ,
1. AGENCY USE ONLY (Leave blank)	2. REPORT DATE	3. REPORT TYPE AN	
	July 2001	F	Final Contractor Report
4. TITLE AND SUBTITLE			5. FUNDING NUMBERS
Short-Term Aging of NeFeB N	Magnets for Stirling Linear A	Alternator Applications	WU-783-82-00-00
6. AUTHOR(S)			NAS3-98008
Janis M. Niedra			NA33-90000
7. PERFORMING ORGANIZATION NAM	E(S) AND ADDRESS(ES)		8. PERFORMING ORGANIZATION
D Fariancia Commun	T		REPORT NUMBER
Dynacs Engineering Company	, Inc.		E-12806
2001 Aerospace Parkway Brook Park, Ohio 44142			E-12800
BIOOK Fark, Offio 44142			
9. SPONSORING/MONITORING AGENC	Y NAME(S) AND ADDRESS(ES)		10. SPONSORING/MONITORING AGENCY REPORT NUMBER
National Aeronautics and Space	ce Administration		
Washington, DC 20546-0001	Ĺ		NASA CR—2001-210952
11. SUPPLEMENTARY NOTES			
Project Manager, Gene E. Sch Center, organization code 5450	0, 216–433–6117.	l Propulsion Technology	Division, NASA Glenn Research
12a. DISTRIBUTION/AVAILABILITY STA	TEMENT		12b. DISTRIBUTION CODE
Unclassified - Unlimited			
Subject Categories: 20 and 33	Distri	bution: Nonstandard	
Available electronically at <a href="http://glt">http://glt</a>	rs.grc.nasa.gov/GLTRS		
This publication is available from the		Information, 301–621–0390.	
13. ABSTRACT (Maximum 200 words)		,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,	
NeFeB type magnets have bee electric power during long dur	ration space missions. These	type of materials provid	en, linear alternators to generate te the highest energy product commer- temperature stability of magnetic

NeFeB type magnets have been proposed for use in free piston Stirling engine driven, linear alternators to generate electric power during long duration space missions. These type of materials provide the highest energy product commercial magnets, thus minimizing alternator size or mass, but do not provide the high temperature stability of magnetic properties found in the SmCo type magnets. Therefore, to apply the NeFeB type magnets at elevated temperatures to multiyear space missions, their long-term aging characteristics must be determined. This report presents 200 hr aging data for 6 types of NeFeB magnets selected from 3 manufacturers. Aging was performed under vacuum at 150 °C, with a steady demagnetizing field of 5 kOe applied. From the data produced by this short-term aging run, candidate magnet types were selected for a planned 12 000 hr long-term run. Depending on the manufacturer's magnet type, remanence losses observed ranged from 0 to 7 percent, when measured at 120 °C on an established recoil line. Also, intrinsic coercivity losses up to about 4 percent were observed for the M-H curve at 120 °C. In some cases, these coercivity losses were not recoverable by recharge of the magnet, indicating a structural change of the material.

14.	SUBJECT TERMS	15. NUMBER OF PAGES		
	Permanent magnets; Neody	17		
	Coercivity; Remanence	2 2,	16. PRICE CODE	
	Coefficient, remainemen			
17.	SECURITY CLASSIFICATION	20. LIMITATION OF ABSTRACT		
	OF REPORT			
	Unclassified			